

202	N66-83030	
Ξ	(ACCESSION NUMBER)	(THRU)
Y FORM		_ none
\CILITY	(PAGES)	(CODE)
ĒΨ	CR 54615	
	(NASA CR OR TMX OR AD NUMBER)	(CATEGORY)



GIIIIIID

GENERAL DYNAMICS

-ASTRONAUTICS

NOW COMBINED WITH CONVAIR

A2136-1 (REV. 6-61)

MECHANICAL PROPERTIES OF Ti-8Al-1Mo-1V ALLOY AT ROOM AND CRYOGENIC TEMPERATURES.

MRG 246

August 2, 1961

PREPARED BY: J.L. CHRISTIAN

GENERAL DYNAMICS/CONVAIR

PAGE

2 August 1961

SUBJECT:

Mechanical Properties of Ti-8Al-1Mo-1V Alloy at Room and Cryogenic Temperatures.

APSTRACT:

The tensile (Fty, Ftu and elongation), weld tensile (Ftu, elongation and joint efficiency) and notched tensile strengths and notched/unnotched tensile ratios were determined at 780, -100°, -320° and -423°F on titanium 8Al-1Mo-1V sheet material. The alloy was tested in the mill-annealed condition. The data obtained show a continuous increase in the yield strength, tensile strength and held tensile strength with reduction in testing temperature. Weld ident efficiencies were 100% at all testing temperatures. Elongations of base metal and weld joints were nearly the same at 78°, -100° and -320°F but decreased sharply at -423°F. Notched tensile strengths and notched/unnotched tensile ratios indicate adequate toughness for structural applications at 78°, -100°, and -320°F. However, the decrease in notched tensile strength at -423°F and the resulting low notched/unnotched tensile ratio indicate poor toughness at -423°F. This alloy is not recommended for structural applications at liquid hydrogen temperatures (-423°F).

SEP 7 1965 CENTAUR PROJECT OFFICE

Prepared By

L. Christian

Approved By

A. Hurlich

Research Group Engineer Materials Research Group

AH:JLC:bts

DISTRIBUTION

	V.	F. Radcliff	595-0
		A. Ehricke	100-0
	G.	D. Davis	510-1
	R.	S. Shorey	597-3
	A.	Hausrath	597-3
	D.	Collins	541-1
	L.	Munson	534-0
	C.	Pruckner	54 1-3
	Н.	Steele	59 3-1
	K.	Hogeland	593-1
	J.	Comber	541 -3
	W.	Gross	563-1
	٧.	A. Babits	592-0
	E.	R. Foor	290-2
- 1	A.	E. Miken	290-3
	4	E. O'Brien	567-3
		the state of the s	

TOE

Distribution

FROM:

Materials Research Group, 592-1

SUBJECT:

Mechanical Properties of Ti-8Al-1Mo-1V Alloy at Room and Cryogenic

Temperatures.

INTRODUCTION

The mechanical properties of a large number of titanium alloys have been determined at cryogenic temperatures. Report MRC-189, dated 14 October 1960, gives the mechanical properties of nine titanium alloys at cryogenic temperatures. Of these alloys, only one, the Ti-5Al-2.5Sn material, retains sufficient toughness for structural applications at -423°F (boiling point of liquid hydrogen). The titanium-8Al-1Mo-1V alloy has higher tensile and yield strengths at 78°F and is more easily rolled into thin gauge sheet material than the Ti-5Al-2.5Sn material. It was therefore the purpose of this investigation to determine if the Ti-8Al-1Mo-1V alloy would be suitable for use at cryogenic temperatures.

MATERIALS

The Ti-8Al-1Mo-1V alloy used in this investigation was supplied by Titanium Metals Corporation of America, in 0.096 inch sheet, heat number M-9519. Table 1 presents the chemical analysis and properties as supplied by certificate of test from the supplier. The material was tested in the as received condition.

PROCEDURE

Blanks for tensile specimens, $9^n \times 1\frac{1}{2}^n$, were identified and sheared in directions both longitudinal and transverse to the direction of rolling. Panels of the alloy were inert-arc fusion welded on production equipment and sheared into tensile blanks. Smooth and welded (EMG-D-1) and notched (MRG-D-10, Notch "A") tensile specimens were machined. A minimum of three tensile tests in the longitudinal and two tests in the transverse directions were performed on both smooth and notched specimens at room temperature (78°F), -100°F (alcohol and dry ice), -320°F (liquid nitrogen), and -423°F (liquid hydrogen). Strain measurements were made by use of extensometers (cryo-extensometer at low temperatures) and a continuous stress strain recorder. Total elongations on both base metal and welds were determined over a 2" gauge length made by scribe marks with a precision block and read under 10X magnification. Strain rates were maintained at 0.002#/"/min. until 0.2% offset yield and then 0.15"/min. until fracture. The 50,000# Baldwin testing machine, strain recorder, strain pacer, and extensometers are periodically checked and approved by standards laboratory.

PAGE TO THE RESERVE THE PAGE TO THE PAGE T

RESULTS AND DISCUSSION

The mechanical properties of Ti-8Al-1Mo-1V at ambient and cryogenic temperatures are given in Table 2. The room temperature properties agree quite well with those obtained by the supplier (see Table 1). Yield and tensile strengths of base metal and tensile strengths of butt welded joints increased with decrease in testing temperature (about 75% from 78°F to -423°F). Elongations of base metal and welded specimens were nearly the same at 78°, -100°, and -320°F but decreased significantly at -423°F. Weld joint efficiencies were 100% at all testing temperatures. Notched tensile strengths increased from 78°F to -320°F but decreased at -423°F. Notched/unnotched tensile retios decreased with reduction in testing temperatures.

Based on the data obtained in this investigation it is believed that the Ti-8Al-1Mo-1V alloy retains adequate toughness for structural use at -320°F, but should not be used for structural applications at liquid hydrogen temperatures (-423°F).

SUMMARY

- 1. Yield and tensile strengths of base metal and tensile strengths of fusion welds incresse about 75% from 78°F to -423°F.
- 2. Elongations of base metal and welds are nearly the same at 78° , -100° and -320° F but decrease at -423° F.
- 3. Weld joint efficiencies are 100% at all testing temperatures.
- 4. Notched tensile strengths and notched/unnotched tensile ratios indicate sufficient toughness for structural applications at 78° , -100° and -3200F but not at -423°F.

TABLE 1

Chemical Composition and Properties of Ti-8Al-1Mo-1V *0.096* Sheet: Mill Annealed 1350°F. 8 Hours: Heat M-9519

Chemistry

Alloying Element	Percent
Aluminum	7.8
Molybdenum	1.1
Vanadium	1.0
Iron	0.09
Carbon	0.02
Nitrogen	0.012
Oxygen	0.08
Hydrogen	0.013

Properties

Direction	Yield Strength	Tensile Strength	Flong.	Pend Test (105° Press Brage)
Long.	137,400 psi	145,400 psi	17%	2.9
Trans.	1 3 5,000 psi	145,100 psi	15.5%	2.9

Certificate of Test from Titanium Metals Corporation of America, Test A-4816, dated August 10, 1960.

100

Included (K ₁ =6.3) Notched (K ₁ =6.3) Tengile Strength (kgi) 156 156 156 156 166 166 176 176	of 11-8A1-190-1V Allo TWCA. Heat # N-9515 Strength Notche 55 55 55 55 55 55 55 55 55 55 55 55 55
	Notched/Unnotched Tensile Ratio 1.09 1.09 1.09 0.98 0.89

GIIIIIID

GENERAL DYNAMICS ASTRONAUTICS